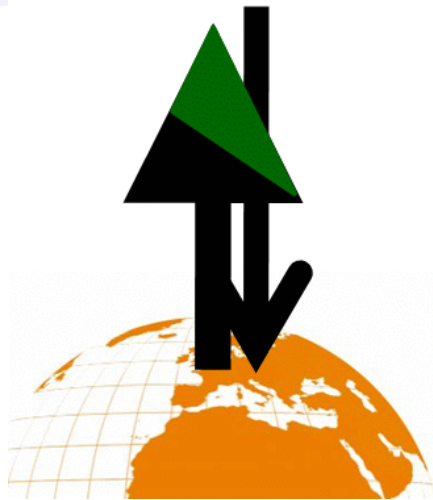


# Payloadproject.com



## Ignition Concept

Contactless ignition by means of  
Electromagnetic waves  
Combination with absorbers

Patent Application:

- 20 Pages
- 9 Claims
- 14 Figures

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## State of the art 2021 - Rocket launches from earth with chem. Rockets

### economic:

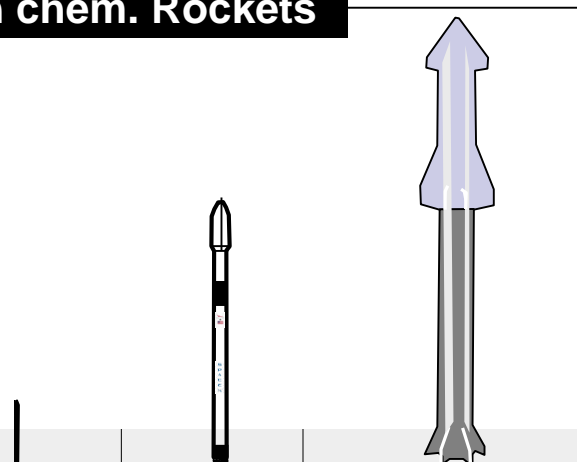
→ Radical innovations necessary to keep up the pace of development and remain competitive

### technical:

→ Payload share still limited!

approx. 1-4% payload for Low Earth Orbit (LEO) - low earth orbit

for smaller rockets tends to be lower



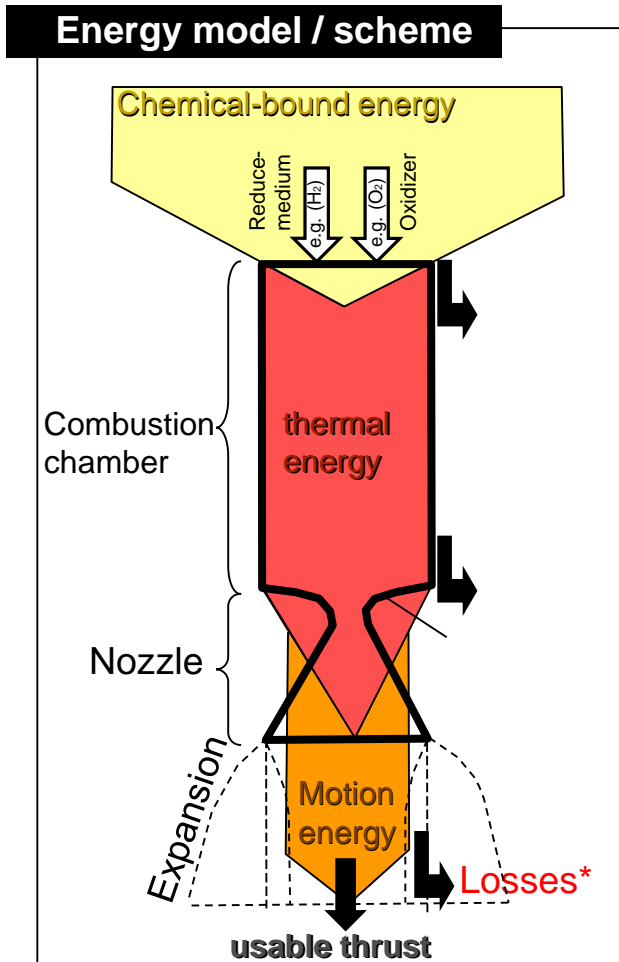
Type	Electron (Rocket Lab)	Falcon 9 (SpaceX)	planned Starship (SpaceX)
Propellant	LOX, RP1	LOX, RP1	LOX, Methan
Take-off mass [t]	13	541	5.000
Payload LEO [t]	0,3	23	>100
Payload LEO [%]	2,3	4,2	ca. 2,0
Cost [US \$ million]	7	62	2
spec. cost [\$/kg]	23.333	2.719	20 (Target: complete recycling)

→ Goal of Lastprojekt.de: more payload share for rockets

## Multiple energy conversions

→ lossy (approx. 40-70% usable thrust[1])

[1] Source: Ernst Messerschmid et al: Raumfahrtssysteme; 4. Auflage, 2011, ISBN 978-3-642-12816-5



**Speed of chem. Conversion limited\*  
(chemical to thermal/motion energy)**

**Principle of thermo-chemistry**

higher combustion chamber temperature  
→ higher exit velocity

→ but\*:

**simultaneously increases:**

**cooling, heat losses/ aggressiveness of the reaction**

**simultaneously decreases: Strength/ life Materials**

Nozzle adapted to specific external pressure → but varies with height above zero\*.

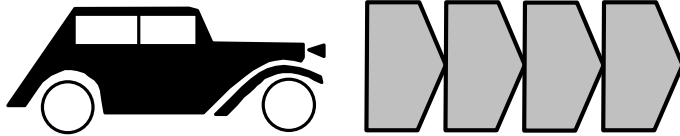
**\* Conclusion: Alternatives to Thermo-Chemistry are advantageous**

## Injection and ignition crucial for high performance

Especially use of microwaves

### Vehicle area - Discontinuous

2 or 4 cycles: Suction, Compression,  
Working, Ejection



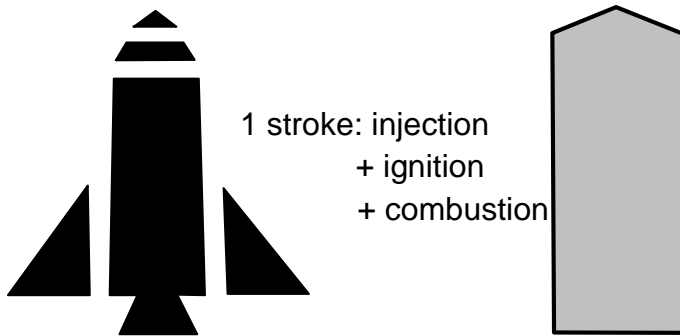
→ Combustion chambers in the liter range  
(~1/1000 m<sup>3</sup>)

**State of the art: advanced**

- Feasibility proven (e.g. MWI AG)
- Fuel savings of up to approx. 30% possible
- According to DE 103 56 916 B3 fuel savings of approx. 66% technically possible
- Microwaves increase flame speed with simultaneous reduction of combustion temperature

### Aerospace - Continuous

1 stroke: injection  
+ ignition  
+ combustion



→ Combustion chambers in the m<sup>3</sup> range  
Throughput rates many times higher

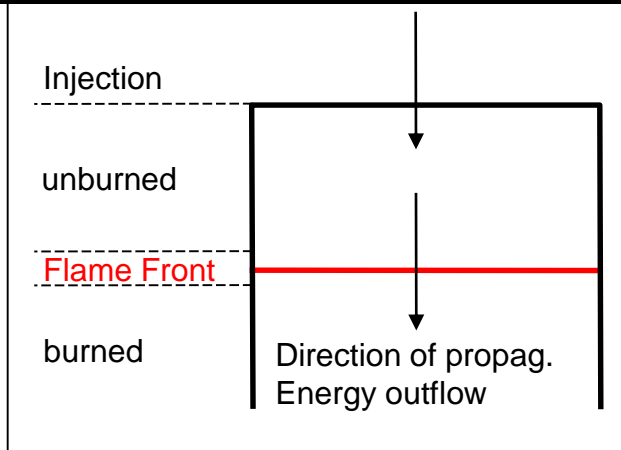
**State of the art: not / approaches available**

- DE3903602A1 inter alia to reduce turbulence
- outside chemical drives: **advanced**
- for electric actuators directed acceleration of supporting agents (e.g. noble gases)
  - Electric propulsion already proven in use!  
(Not suitable for rocket launches from earth)

# Ignition in general (laminar / turbulent)

Dependent on numerous factors (injection rate, pressure, combustion speed and size of ignition range).

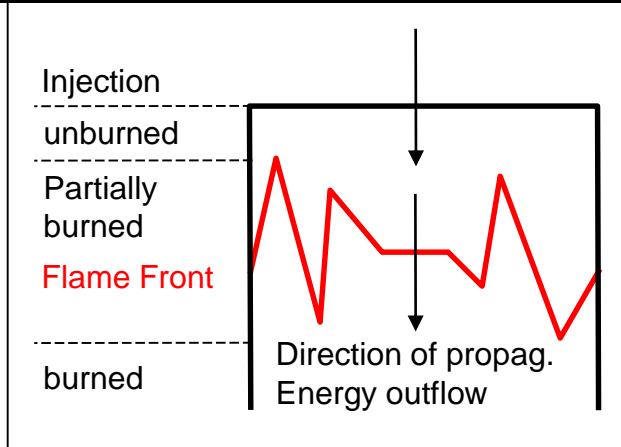
## laminar combustion in comb. chamber



## Properties:

- uniform conversion of chemical to thermal energy
- ideal for thermal power plants
- due to preceding pressure waves and their reflection
- or possible extinction with trailing combustion, however, can result in a high proportion of internal friction (rather disadvantageous) → more heat
- **Conclusion: not necessarily for drives**  
**can have a disadvantageous effect**

## turbulent comb. in combustion chamber



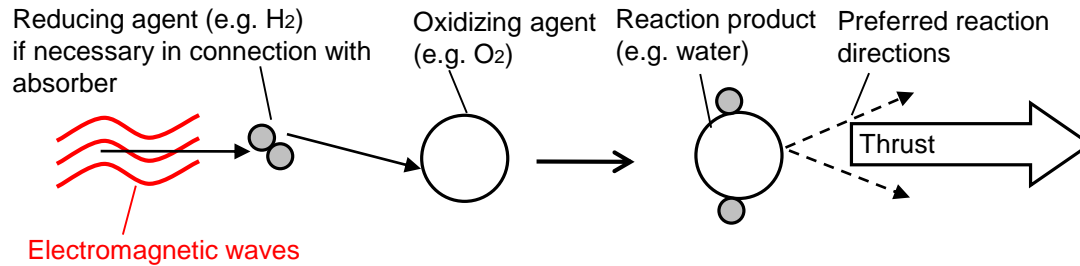
## Properties:

- requires larger ignition range and higher comb. speed
- is partly aimed at in the automotive sector (discont. systems) due to better conversion of chemical into kinetic energy with closed combustion chamber
- Pressure surges should remain limited to avoid engine damage
- **Conclusion: modified change of state**  
**for propulsion systems partly advantageous**

## Aim: as directed ignition as possible in the direction of thrust

Electromagnetic waves: e.g. microwaves, radio waves, (X-ray waves), magnetic waves

### Scheme of a reaction



**Ceramics permeable for electromagnetic waves**  
→ Coupling into comb. chamber

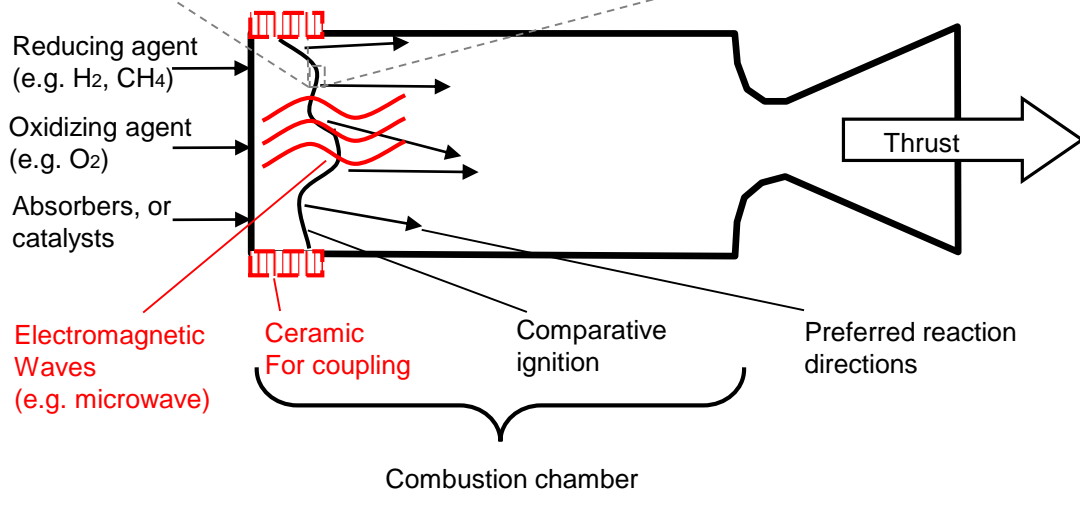
**Electrotechnical supply possible from:**

- generator at turbopump
- electromagnetic generator on the outside of the engine
- thermocouple e.g. at combustion chamber

**Reduction of required electrical power by:**

- Absorbers for EM waves
- **e.g. catalysts** (reduction of activation energy)
- additional metal particles (triorgol systems)

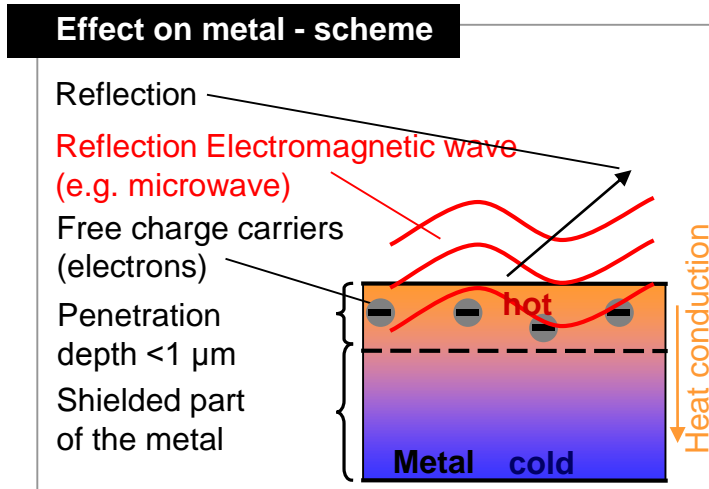
### Ignition concept on rocket engine



Electromagnetic waves can penetrate certain substances

Goal: couple energy into fuel as quickly + highly conc. as possible

→ Saving energy and improving ignition



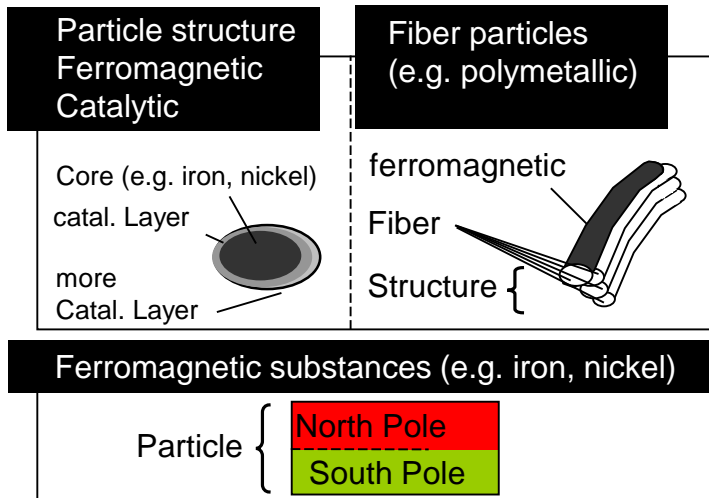
Electromagnetic couplability depending on dielectric constant  
 e.g.  $\text{H}_2\text{O}$  polar (approx. 80x higher dielectric const. than  $\text{H}_2 / \text{O}_2$ )  
 → Household microwave specifically heats water molecules

fuel has small dielectric constant → hardly any coupling  
 However, changed / improved under pressure and temperature

Microwaves: are generally reflected by metals.

→ **However, the following effects are beneficial for metals:**

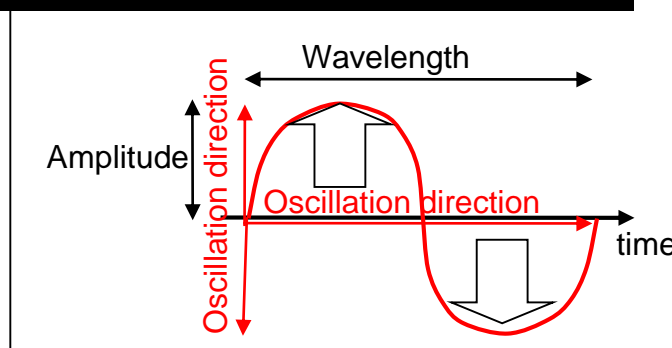
- microwaves can penetrate approx.  $< 1 \mu\text{m}$  (free charges)
  - heated metal parts are ionized / polarized
  - reflections can lead to resonances → heat
  - dissolved metal parts can lead to polarizations
  - polymetallic compounds with polarized bonds
  - Alloys / dopants / surfaces
  - Combination with other electromagnetic waves
- **Metals high couplability possible  $< 1 \mu\text{m}$**   
 → **Ferrites / ferromagnets are good absorbers**  
 + **catalysts lower activation energy**



# Ignition - directed combustion

Approach: the more aligned combustion - the higher combustion speed and the lower temperature

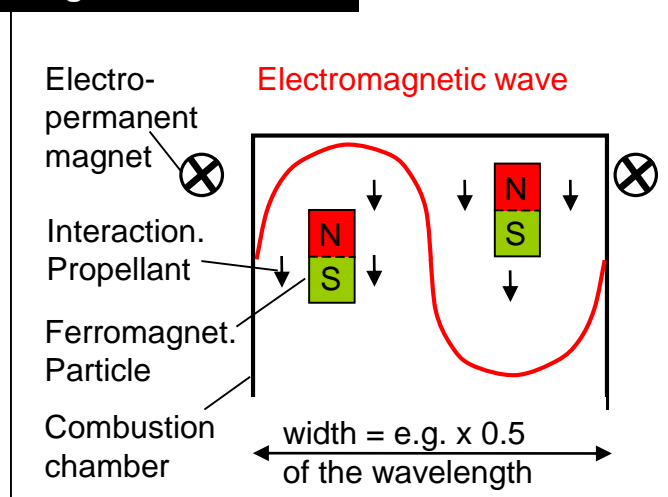
Electromagnetic wave = "transversal" (perpendicular)



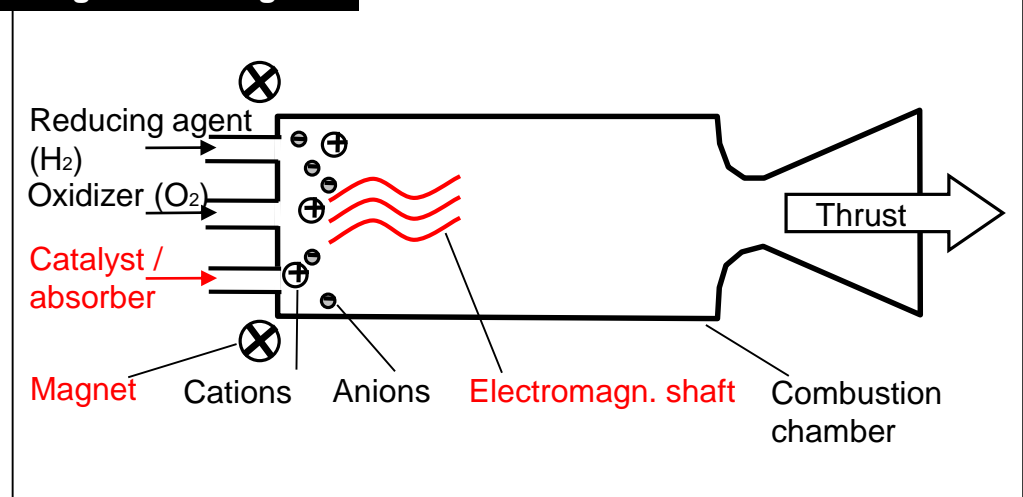
## Ignition and alignment via

- Selection of suitable wavelength to combustion chamber
- magnets for ionization / polarization propellant
- if necessary additional ionization radiation
- metallic particles
- if necessary ferromagnetic particles + magnetic field
- targeted excitation / thermal stimulation from one side

## Alignment - scheme



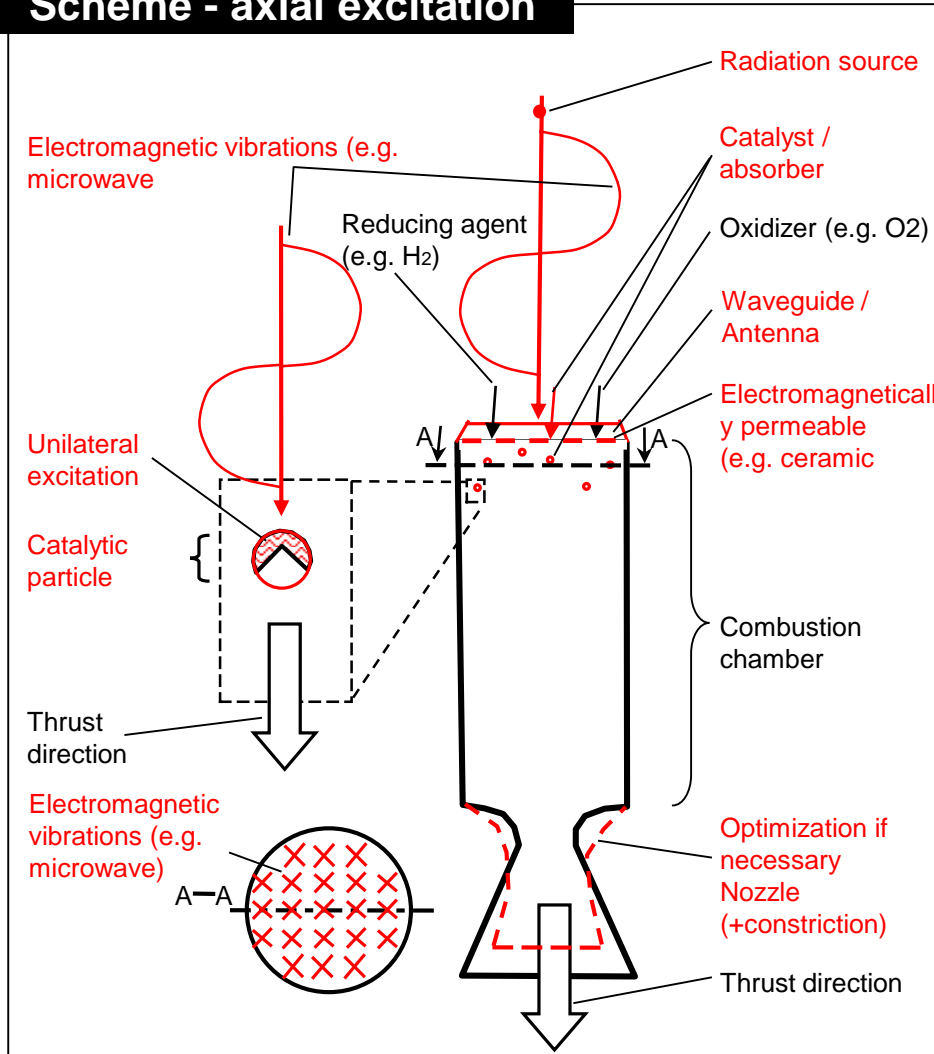
## Alignment - Engine





## Utilization of the propagation speed

### Scheme - axial excitation



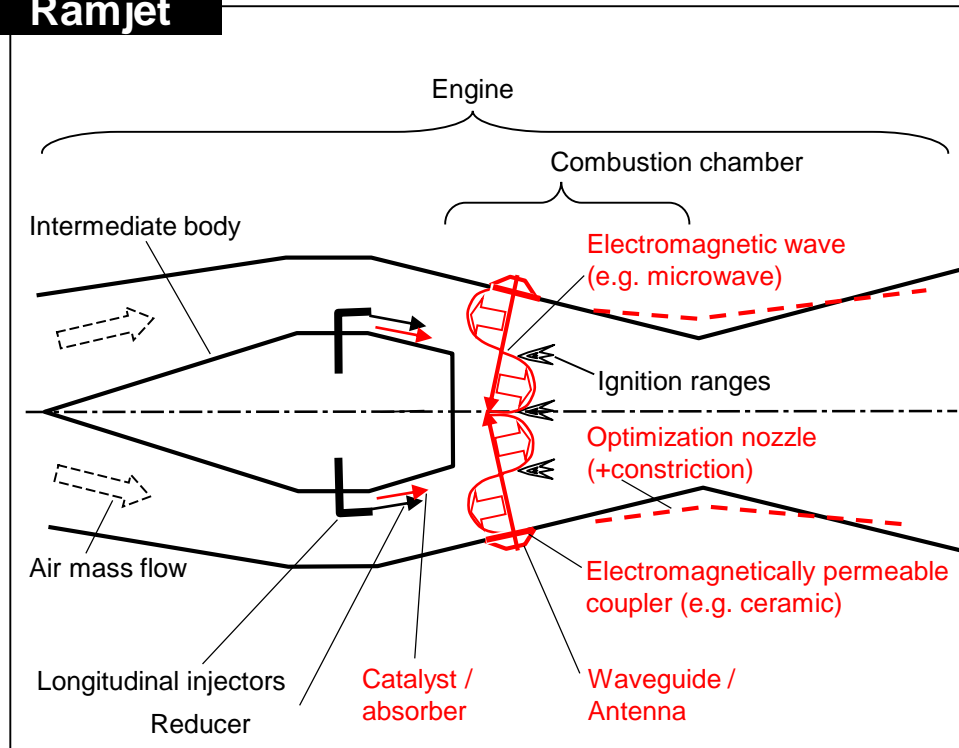
### Desired properties:

- comparative space ignition
- if necessary reflection of electromagnetic waves at catalysts / absorbers
- Uniform heating takes place in the direction of flow
  - gradual heating
  - longer time for absorption energy
  - Potentially less electrical power required for microwave
- Targeted concentrated heating of catalysts / absorbers
  - More reaction on the reverse side
  - Acceleration in engine direction
  - Increase in combustion rate
- Other areas are cooled down as catalyst lowers activation energy

# Ignition from outside possible via ceramic couplers

→ Principle also applicable for scramjet

## Ramjet



## targeted properties:

- Ignition from the side (alternatively ignition from the intermediate body)
- Space ignition / large ignition range
  - higher burnout
  - lower consumption
  - less turbulence
- Elimination of igniters / ignition holders
  - Avoidance of aerodynamic resistance
- Microwaves are more likely to lower temperature:
  - less cooling
  - improved heat resistance
  - higher burn rate
- Constriction can be optimized

# Goal: Alternative to thermo-chemistry

